

## **Discussions on the influence of residual stresses to the fatigue of layered polymer composites**

Institute of Lightweight Engineering and Polymer Technology (ILK)

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## *“Evaluation and modelling of the fatigue damage behaviour of polymer composites at reversed cyclic loading”*

### Objectives

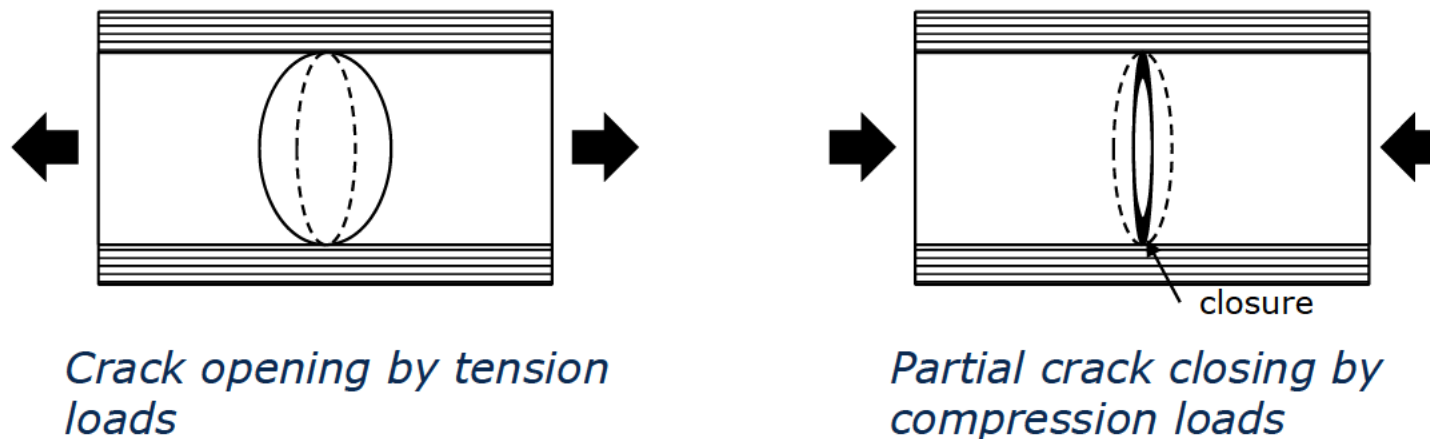
- Analysis of damage behaviour of CFRP under tension and compression on fibre-matrix-, ply- and laminate-level
- Experimental and numerical characterisation of tension-compression asymmetry
- Influence of tension-induced delaminations on compression behaviour
- Mathematical description of the effects of load reversal, block sequence and block length on the damage process
- Extension of existing FEM damage model



# Objectives of the presented work



- Understanding crack opening due to residual stresses
  - Controllable model environment (cross ply laminate)
  - Measuring residual stresses from COD
- Measurement of local strain field between adjacent cracks
  - Estimation of crack opening from strain measurements
  - Influence of load reversal on deformation behaviour



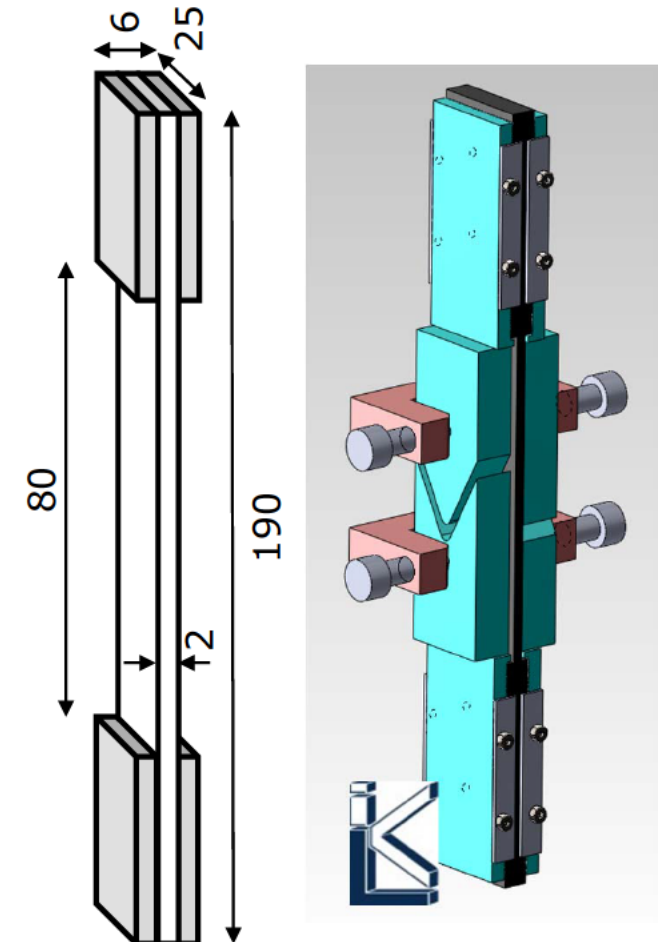
# Materials



- Fibre: T700SC carbon fibre (6k)
- Resin: Araldite LY556/HY917/DY070
- Lay-up:  $[0_2/90_7/0_2]$
- Process: Filament winding + RTM
- Specimen geometry:
  - 190 x 25 x 2 mm (L x W x H)
  - 2 mm aluminium end tabs, extensometer
  - Ply thickness  $t_{nom} = 0.181$  mm

Elastic properties			
	Mean	Std. dev.	Units
$E_{11}$	129430	5445	MPa
$E_{22}$	8049	437	MPa
$E_{33}$	8049	437	MPa
$\nu_{12}$	0.317	0.00763	-
$\nu_{13}$	0.317	0.00763	-
$\nu_{23}$	0.41	-	-
$G_{12}$	3916	194.05	MPa
$G_{13}$	3916	194.05	MPa
$G_{23}$	2854	154.96	MPa

Strength Properties			
	Mean	Std. dev.	Units
$R_p^{(+)}$	2089	53	MPa
$R_p^{(-)}$	1032	221	MPa
$R_s^{(+)}$	36.2	5.3	MPa
$R_s^{(-)}$	164.4	7.9	MPa
$R_{ps}$	52.2	5.4	MPa

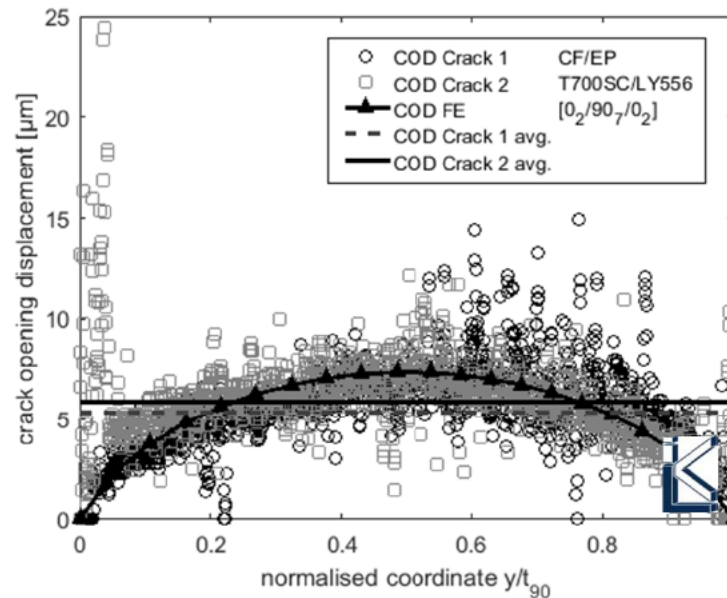


*Specimen geometry and anti-buckling device*

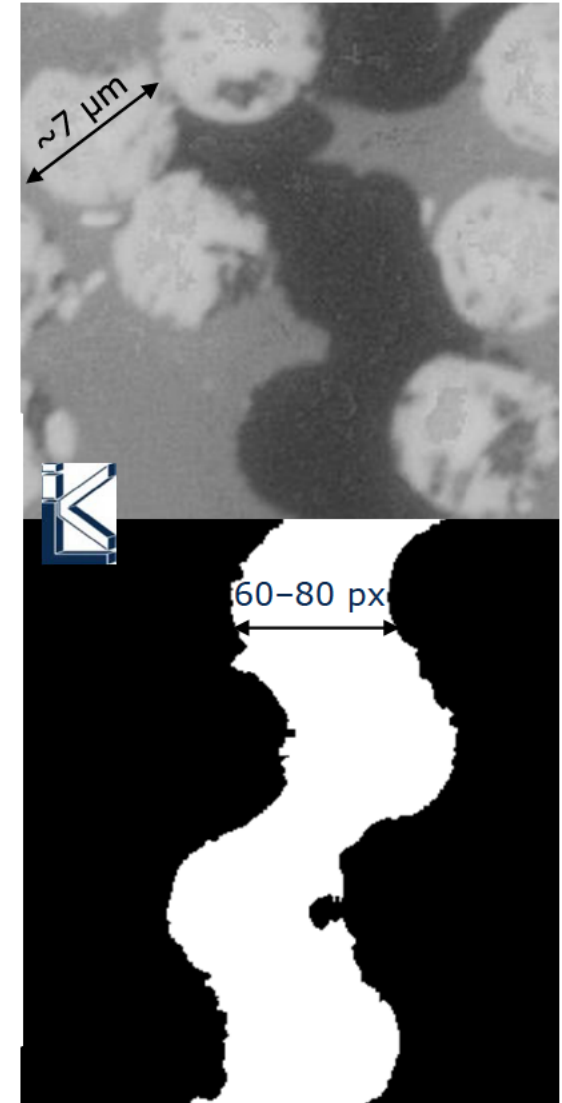
# Residual crack opening



- Two non-interacting cracks after cyclic loading
  - $R = -1$ ,  $\sigma_o^{lam} = 95$  MPa,  $n = 5000$  cycles
- High resolution microscopic images
  - 500-fold, z-stacking, BW-conversion, 60 – 80 pixels between crack faces
  - Row wise summation of white pixels, calibration
- Comparison to FE-results for calculating residual stresses



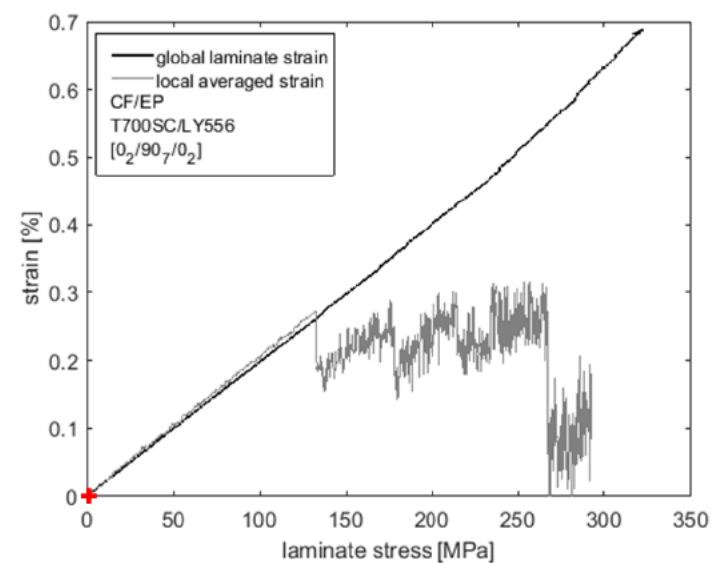
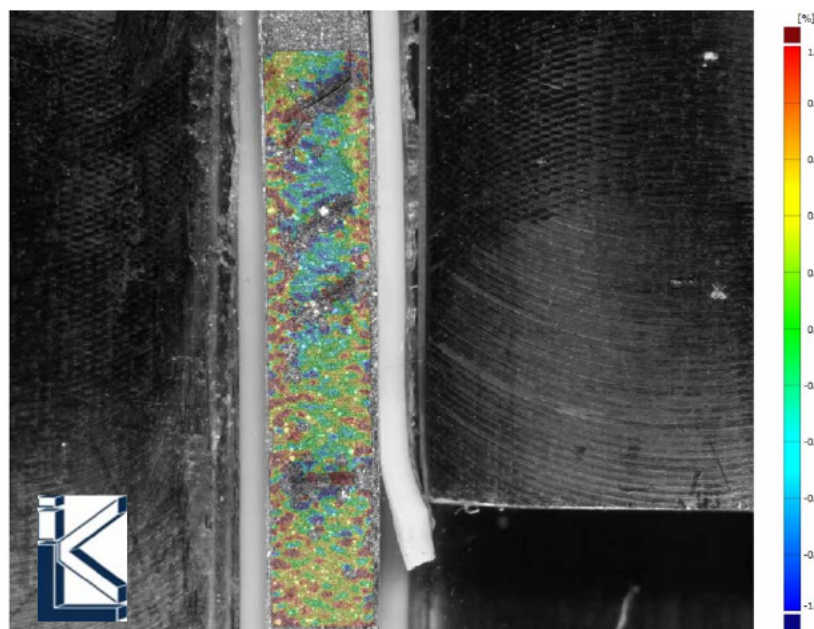
$$\sigma_{res,2}^{90} = 25.7 \text{ MPa @ } \Delta T = 110 \text{ K}$$



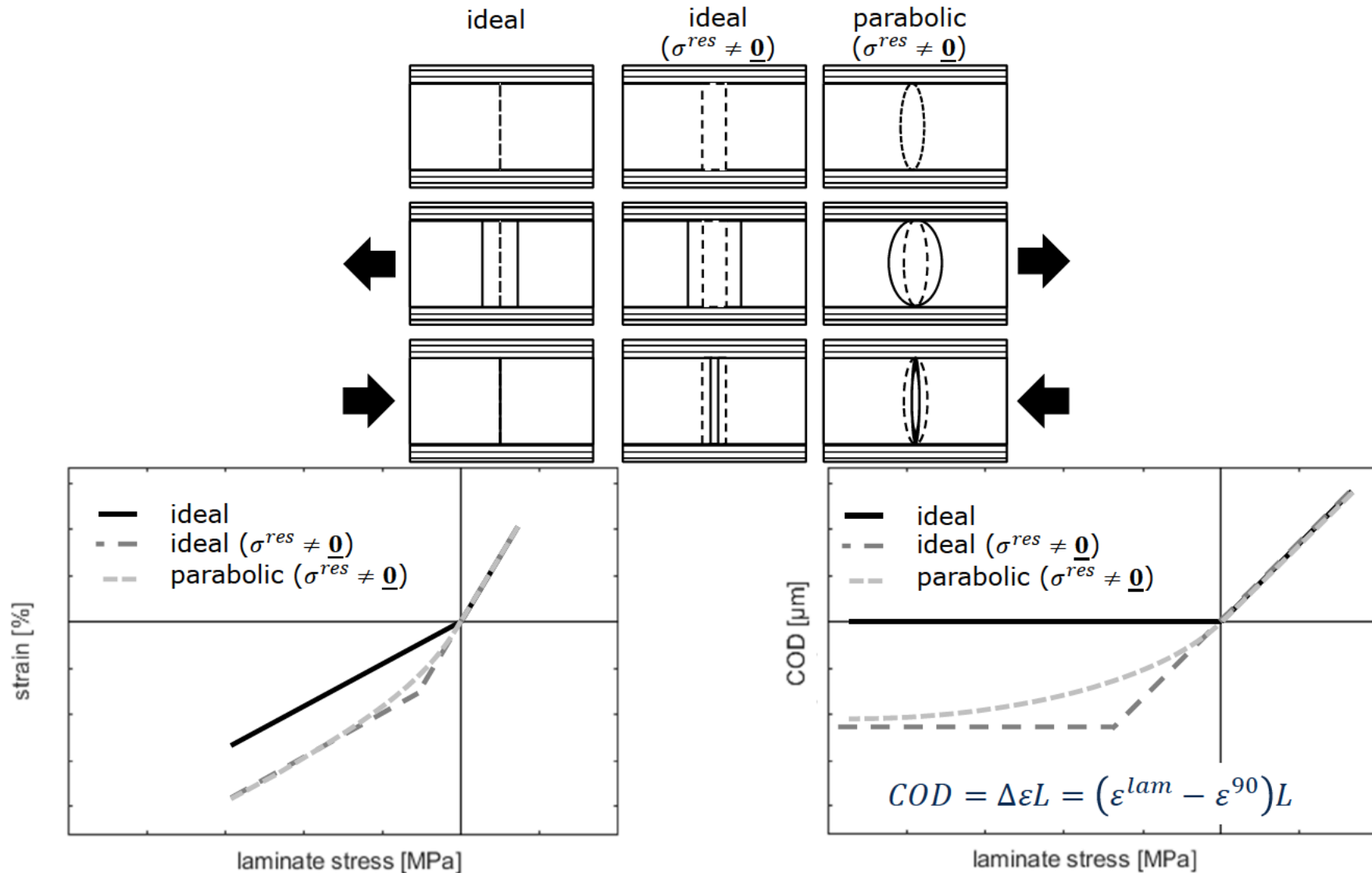


# Analysis of local strain field

1. Yielding transverse cracks by static tension load
  - Relaxation of strains in 90°-ply
  - Cracks form where strains are recovered
2. Cracked specimen loaded in tension and compression
  - $R = \{0; -1; -2.66\}$
  - $\sigma_o = 95$  MPa

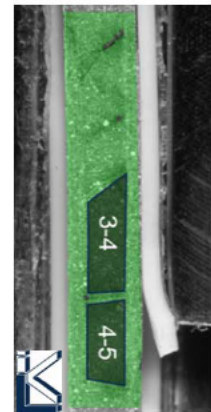
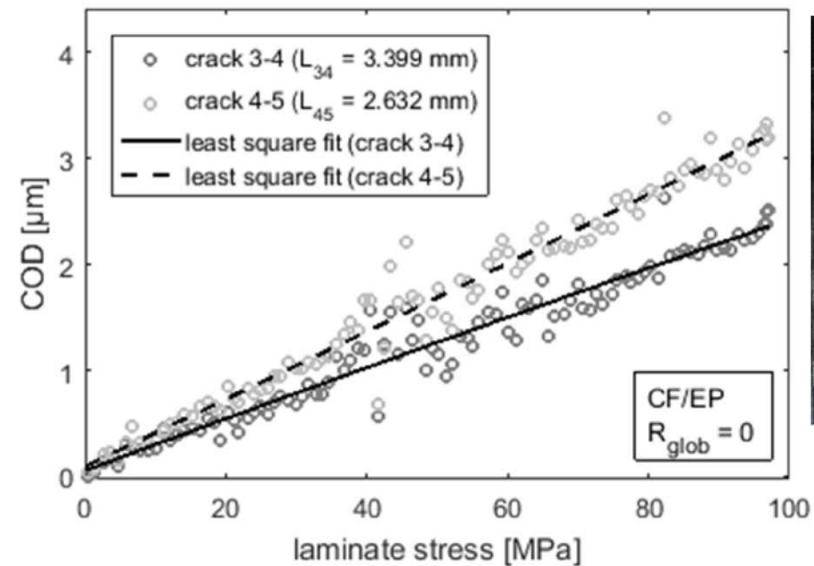
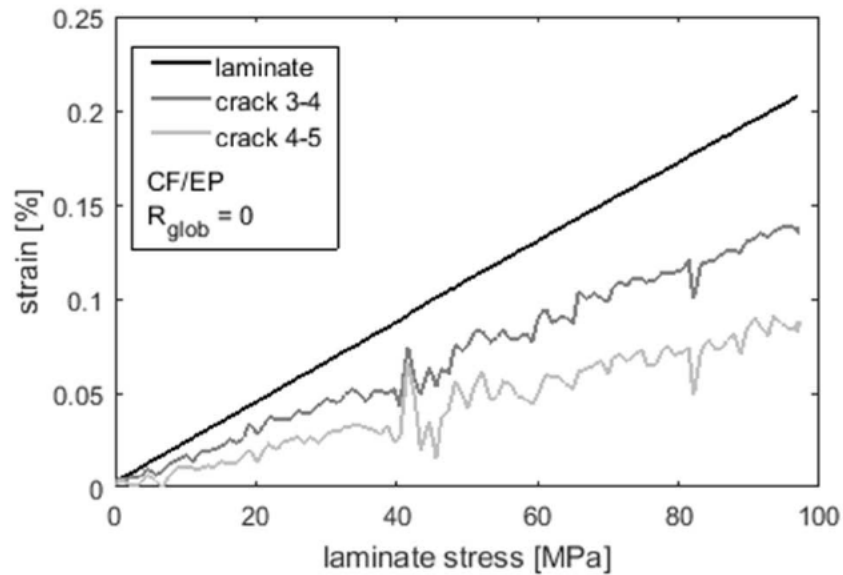


# Estimations for load reversals



# COD under load reversal

Tension-tension ( $R = 0, \sigma_o = 95 \text{ MPa}$ )

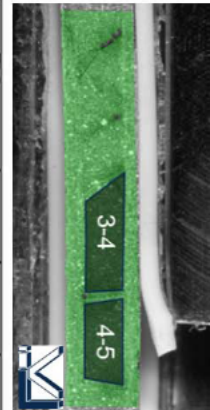
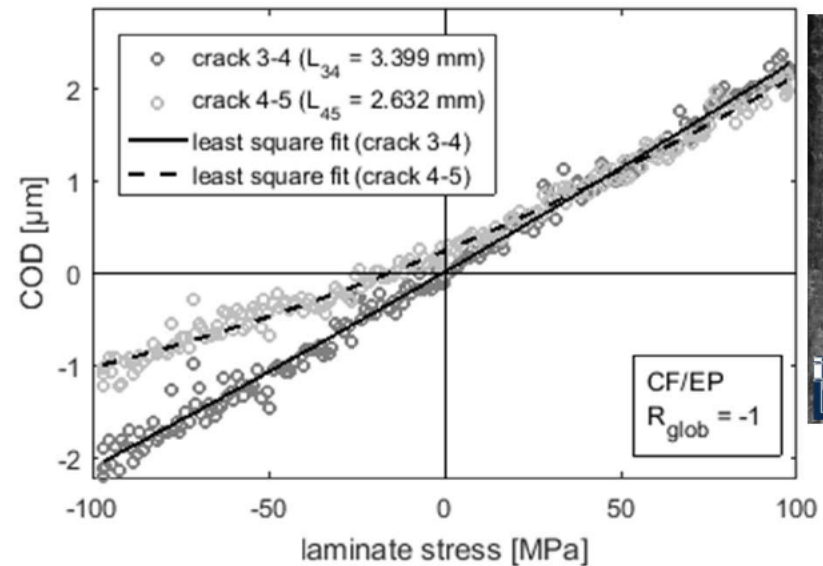
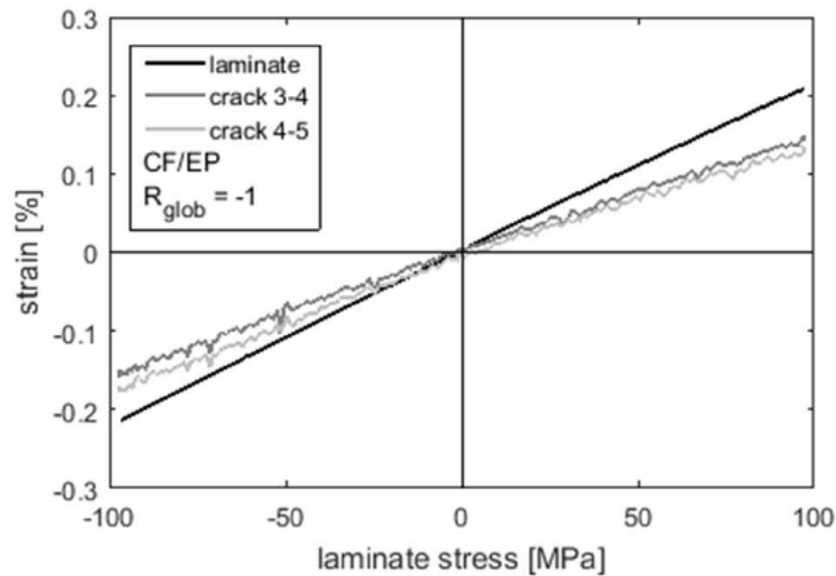


$$COD = \Delta\epsilon L = (\epsilon^{lam} - \epsilon^{90})L$$



# COD under load reversal

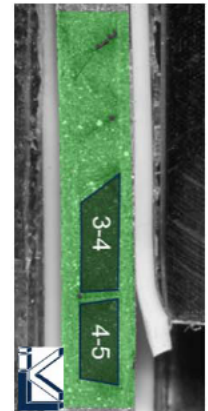
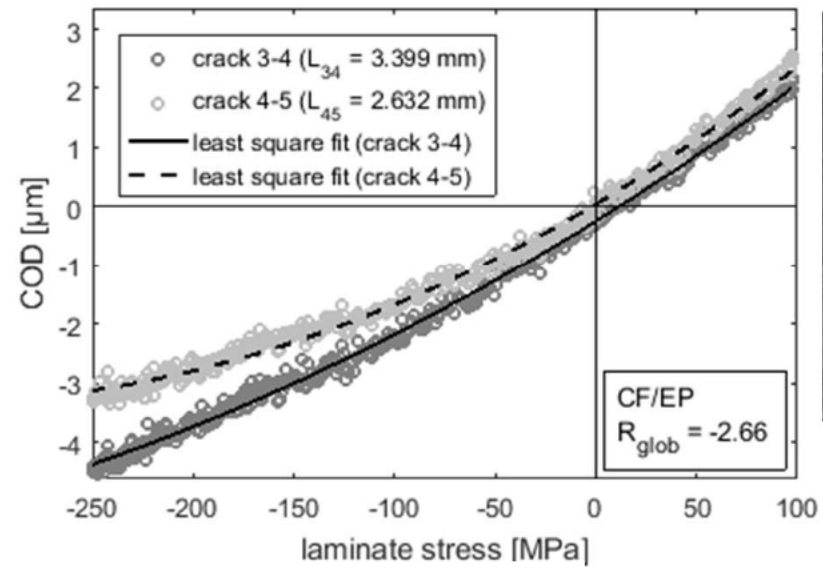
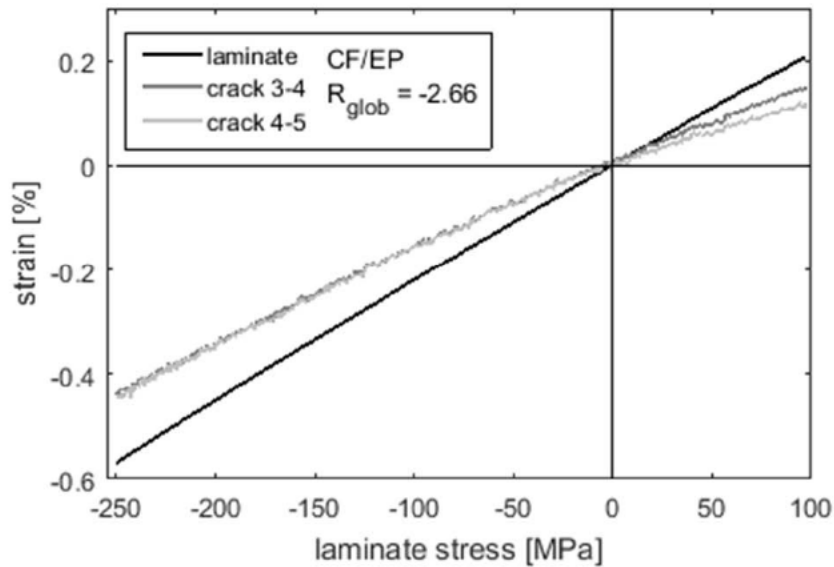
Tension-compression ( $R = -1, \sigma_o = 95 \text{ MPa}$ )



$$COD = \Delta \varepsilon L = (\varepsilon^{lam} - \varepsilon^{90})L$$

# COD under load reversal

Tension-compression ( $R = -2,66, \sigma_o = 95 \text{ MPa}$ )



$$COD = \Delta\epsilon L = (\epsilon^{lam} - \epsilon^{90})L$$

# Conclusion



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- Transverse cracks remain open in case of residual tension stresses
  - Layerwise residual stress evaluation possible
  - Cracks open under tension and close gradually under compression
    - source for nonlinearities and local change of stress ratio
  - Comprehensive fatigue modelling strategies should consider:
    - Residual stress state
    - Physical damage behaviour (cracking, delamination)
    - Damage dependent tension-compression asymmetry

## Outlook

- Comprehensive understanding of stress-strain-hysteresis
- Quantification of stress ratio effects and incorporation to existing FE-based damage models