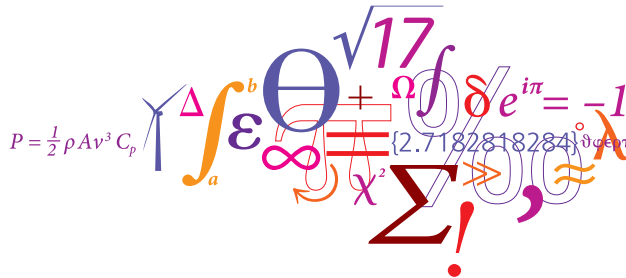


# Micro-scale damage from cyclic bending loads observed in UD Basalt/Epoxy composites

Ulrich A. Mortensen

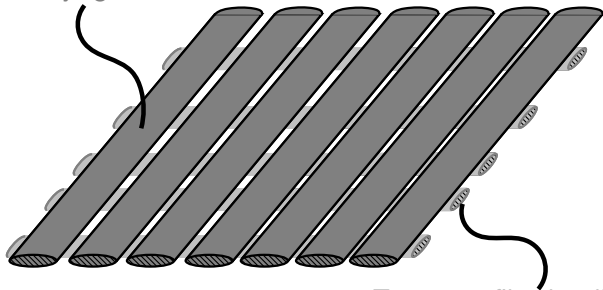
Technical University of Denmark

*Department of Wind Energy , Section of Composites & Materials Mechanics*



## Fibre architecture of Non-crimp fabric

Load Carrying fibre bundles



Transverse fibre bundles

Top side - Longitudinal



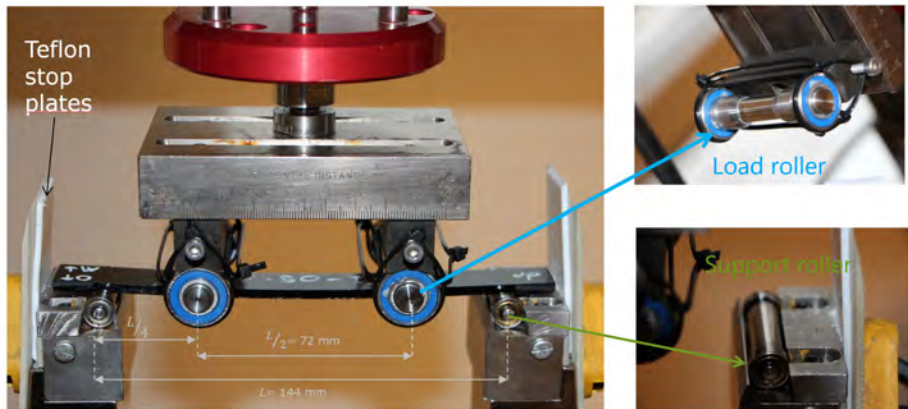
Bottom side - Transverse

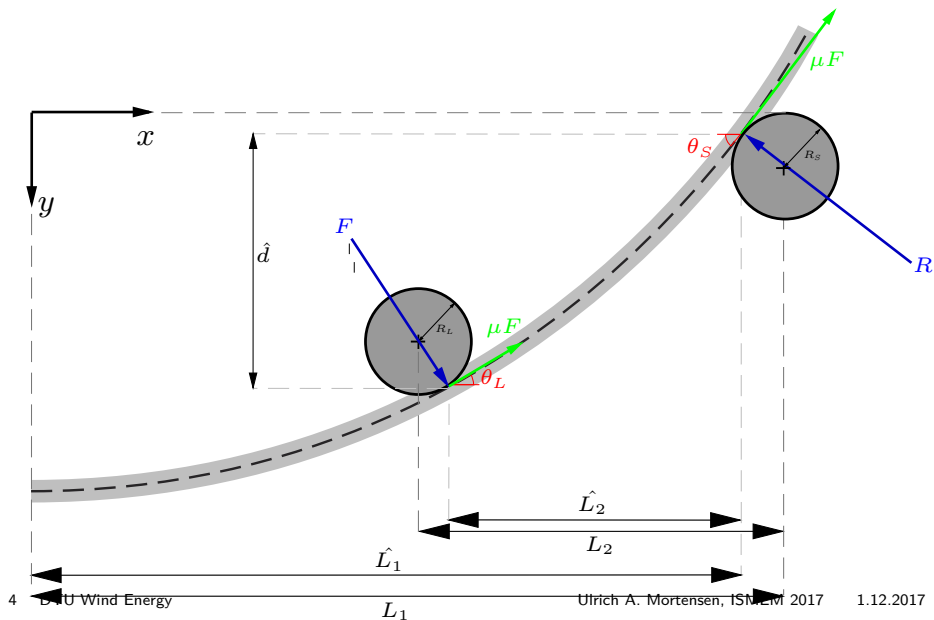


- Transverse (backing bundles and sewing thread) bundles are needed for handling (also in Wind Turbine Industry)

# Specimen Scale

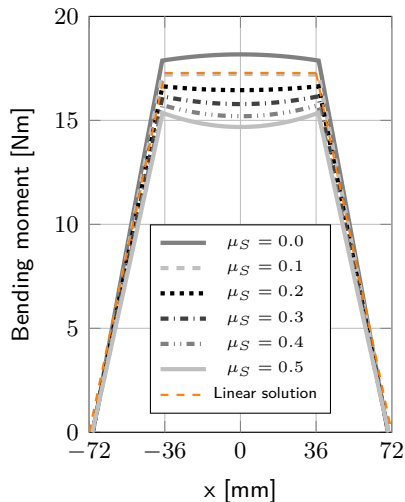
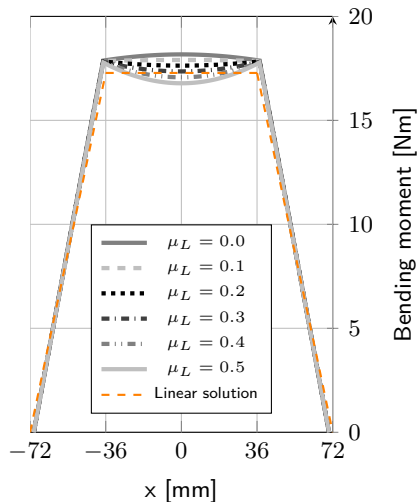
## Set-up and specimen shape





$$M = \frac{P}{2} \left[ \hat{L}_1 - x + y \underbrace{\frac{\tan \theta_s - \mu_s}{1 + \mu_s \tan \theta_s}}_{\text{Non-linear term}} \right] \quad \text{for } x \in [\hat{L}_1 - \hat{L}_2 ; \hat{L}_1] \quad (1)$$

$$M = \frac{P}{2} \left[ \hat{L}_2 + y \underbrace{\frac{\tan \theta_s - \mu_s}{1 + \mu_s \tan \theta_s} - (y - \hat{d}) \frac{\tan \theta_L + \mu_L}{1 - \mu_L \tan \theta_L}}_{\text{Non-linear terms}} \right] \quad \text{for } x \in [0 ; \hat{L}_1 - \hat{L}_2] \quad (2)$$

(a)  $\mu_S \in [0.0; 0.5]$  and  $\mu_L = 0.0$ (b)  $\mu_L \in [0.0; 0.5]$  and  $\mu_S = 0.0$

# Specimen Scale S-N Curve

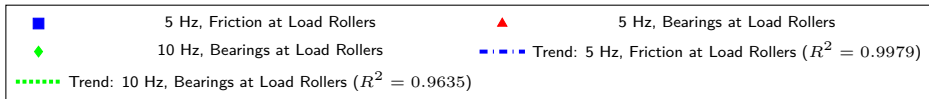
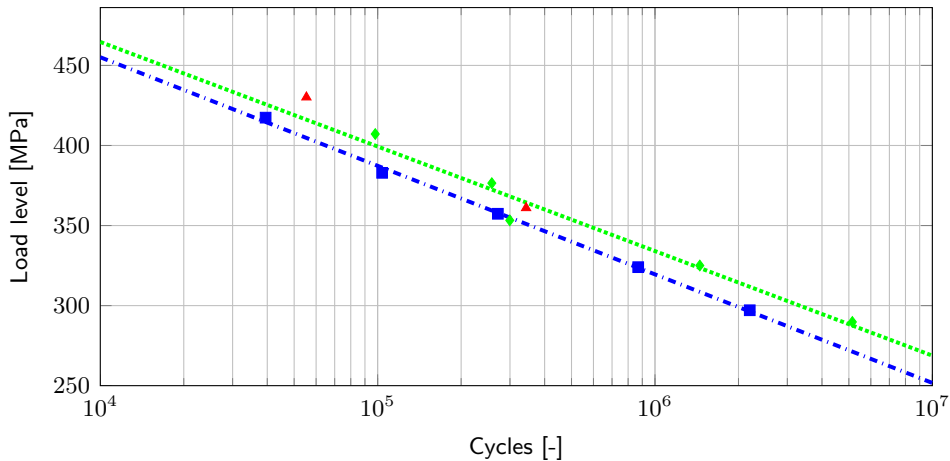
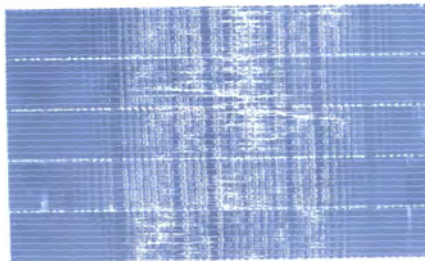




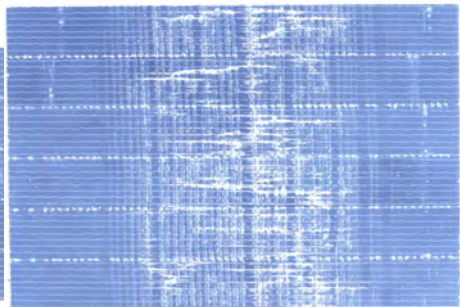
Figure Removed. Will Be  
available in paper to be  
published in 2018



# Macro scale damage due to patterns of transverse fibre bundles



**(a)** Batch A, small friction at load rollers

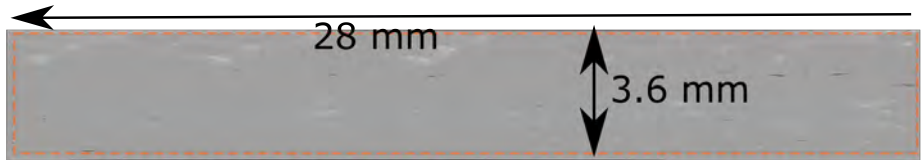


**(b)** Batch B, No friction at load rollers

Micro-scale  
**Large Field of View Microscopy**

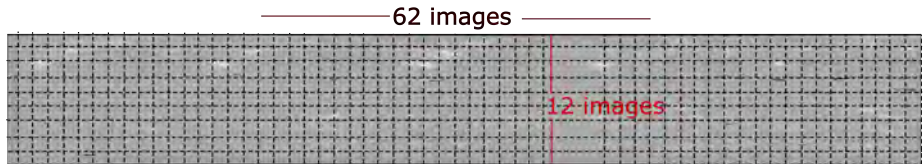


**Figure:** Large Field of View Microscopy. 28 mm by 3.6 mm captured with 700-800 images.



**Figure:** Large Field of View Microscopy. 28 mm by 3.6 mm captured with 700-800 images.

Micro-scale  
Large Field of View Microscopy



**Figure:** Large Field of View Microscopy. 28 mm by 3.6 mm captured with 700-800 images.

# Examples of micro-scale matrix cracks and fibre breaks

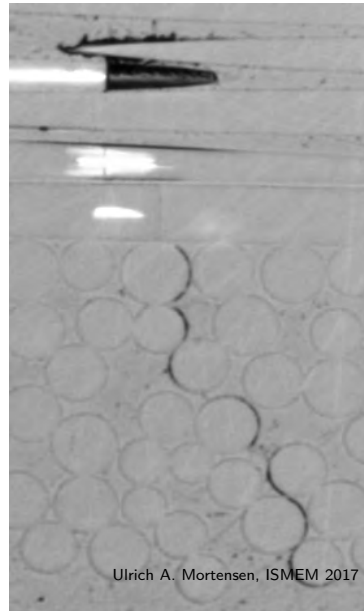
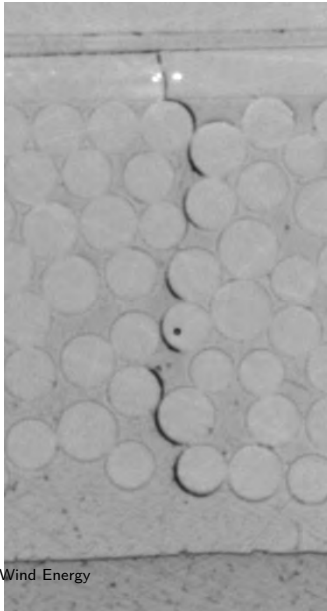
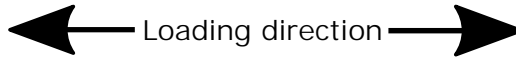
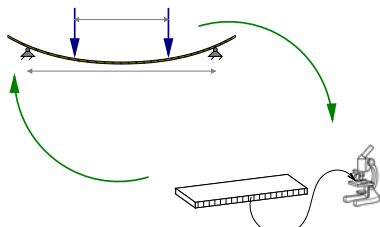




Figure Removed. Will Be available in paper to be published in 2018



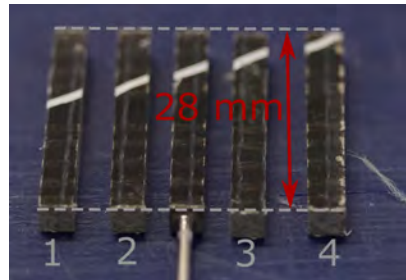
- Size: 9.7 mm by 3.6 mm, 386 images stitched together
- $[1 - 15] \cdot 10^3$  with  $10^3$  cycles per interval
- $[15 - 95] \cdot 10^3$  with  $10^3$  cycles per interval
- $[15 - 95] \cdot 10^3$  with  $10^3$  cycles per interval
- $[95 - 225] \cdot 10^3$  with  $25 \cdot 10^3$  cycles per interval
- $[225 - 1225] \cdot 10^3$  with  $250 \cdot 10^3$  cycles per interval



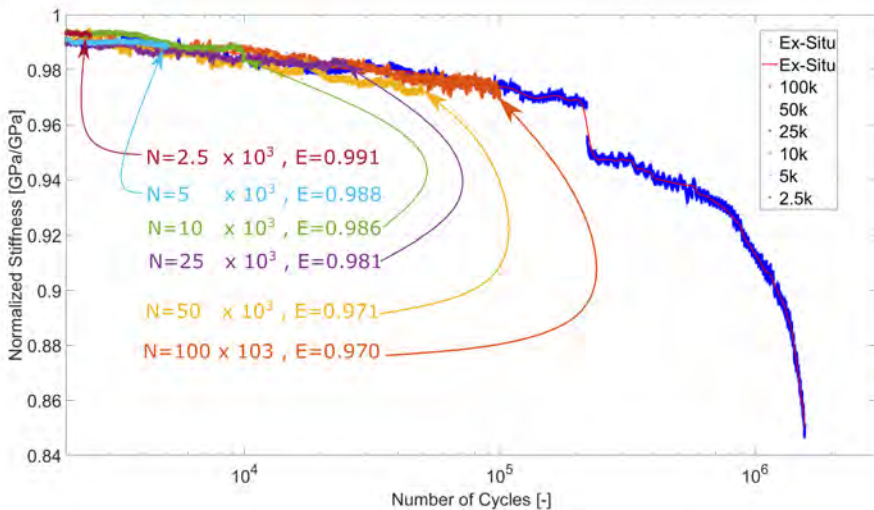


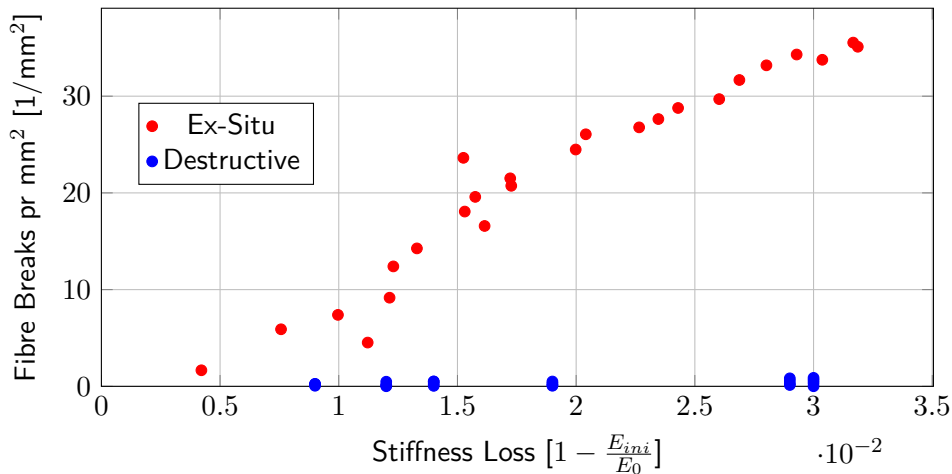
## Micro-scale Destructive Microscopy Test

- Six specimens cut adjacent to each other
- All loaded with 325 MPa maximum stress at surfaces
- No. of loadings  $N = [2.5, 5, 10, 25, 50, 100] \cdot 10^3$

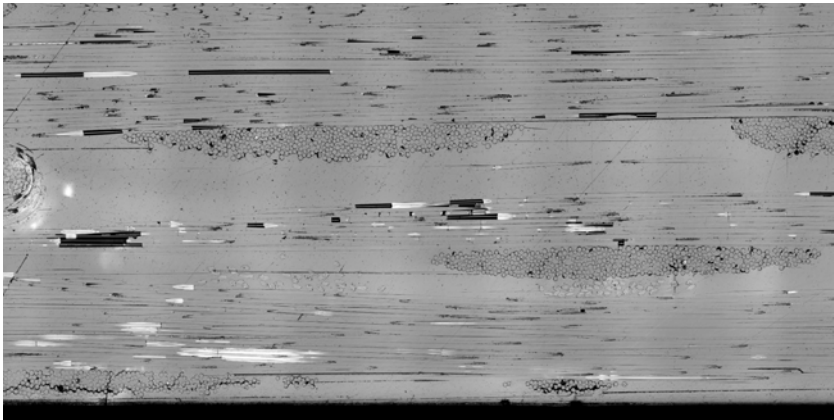


# Micro-scale Stiffness Degradation



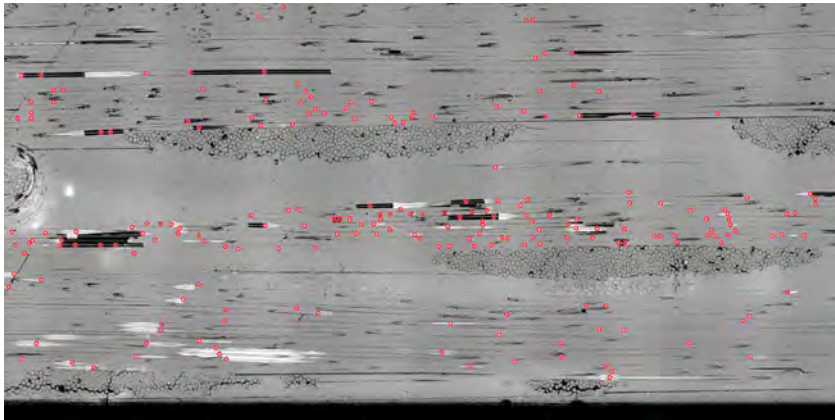


# Distances from Fibre breaks to Transverse fibre bundles - Ex-Situ - $N = 10 \cdot 10^3$



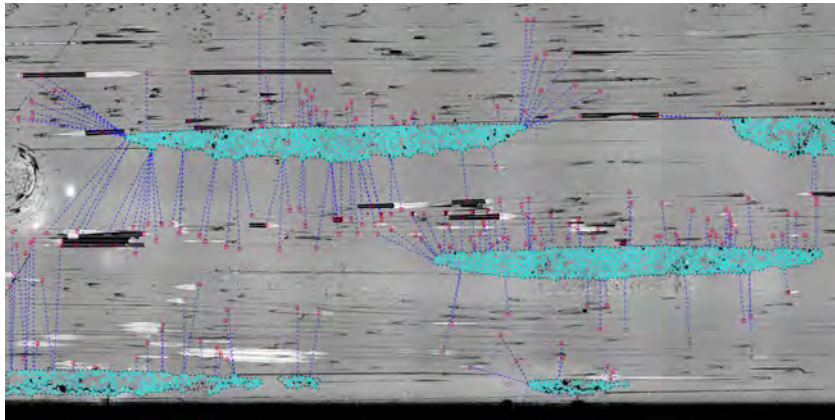
**Figure:** Fibre break positions in Ex-Situ study

# Distances from Fibre breaks to Transverse fibre bundles - Ex-Situ - $N = 10 \cdot 10^3$



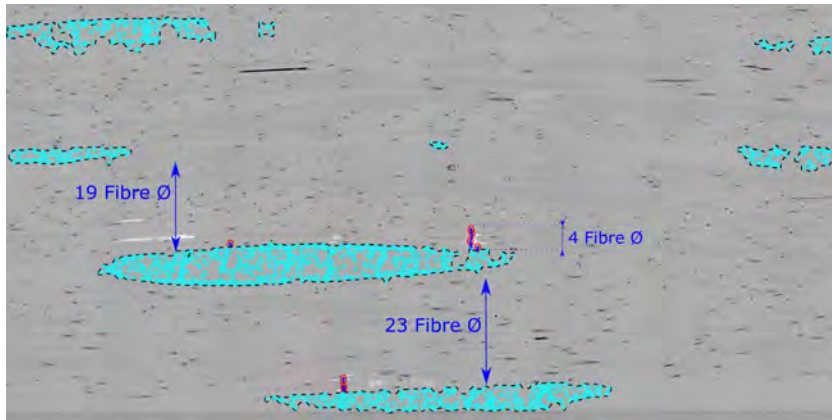
**Figure:** Fibre break positions in Ex-Situ study

# Distances from Fibre breaks to Transverse fibre bundles - Ex-Situ - $N = 10 \cdot 10^3$

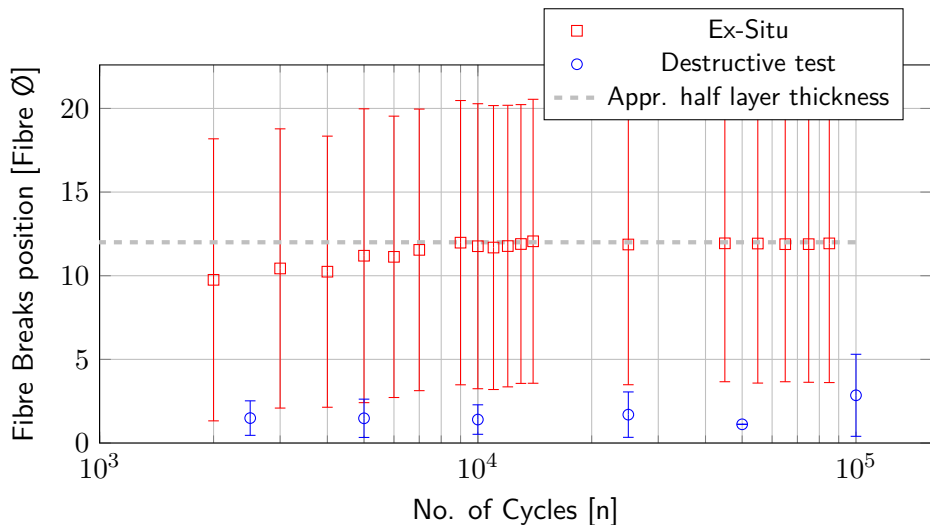


**Figure:** Fibre break positions in Ex-Situ study

# Distances from Fibre breaks to Transverse fibre bundles - Destructive test - $N = 10 \cdot 10^3$



**Figure:** Fibre break positions in Destructive testing





- Edge-effects makes Ex-situ study microscopy studies quantitatively unreliable, but may give some qualitative insight into damage mechanisms
- Transverse Matrix cracks in transverse fibre bundles are the main driver for breakage of fibres in the load carrying fibre bundles in uni-directional non-crimp glass fibre fabric.

Thank You!